



# **Mechanical Design**

## **MECH 330**

1<sup>ST</sup> semester 2021-2022

### **Chapter 6**

## **Fatigue Failure Resulting from Variable Loading**

### **Solved - Problems**

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# **SOLVED PROBLEMS**

## EXAMPLE 6-7 (Textbook)

Consider an unnotched specimen with an endurance limit of 55 kpsi. If the specimen was notched such that  $K_f = 1.6$ , what would be the factor of safety against failure for  $N > 10^6$  cycles at a reversing stress of 30 kpsi?

(a) Solve by reducing  $S'_e$ .

(b) Solve by increasing the applied stress.

### SOLUTION

(a) The endurance limit of the notched specimen is given by

$$S_e = \frac{S'_e}{K_f} = \frac{55}{1.6} = 34.4 \text{ kpsi}$$

and the factor of safety is

$$n = \frac{S_e}{\sigma_a} = \frac{34.4}{30} = 1.15$$

(b) The maximum stress can be written as

$$(\sigma_a)_{\max} = K_f \sigma_a = 1.6(30) = 48.0 \text{ kpsi}$$

and the factor of safety is

$$n = \frac{S'_e}{K_f \sigma_a} = \frac{55}{48} = 1.15$$

## EXAMPLE 6-8 (Textbook)

A 1015 hot-rolled steel bar has been machined to a diameter of 1 in. It is to be placed in reversed axial loading for 70 000 cycles to failure in an operating environment of 550°F. Using ASTM minimum properties, and a reliability of 99 percent, estimate the endurance limit and fatigue strength at 70 000 cycles.

### SOLUTION

From Table A-20,  $S_{ut} = 50$  kpsi at 70°F. Since the rotating-beam specimen endurance limit is not known at room temperature, we determine the ultimate strength at the elevated temperature first, using Table 6-4. From Table 6-4,

$$\left( \frac{S_T}{S_{RT}} \right)_{550^\circ} = \frac{0.995 + 0.963}{2} = 0.979$$

The ultimate strength at 550°F is then

$$(S_{ut})_{550^\circ} = (S_T/S_{RT})_{550^\circ} (S_{ut})_{70^\circ} = 0.979(50) = 49.0 \text{ kpsi}$$

The rotating-beam specimen endurance limit at 550°F is then estimated from Eq. (6-8) as

$$S'_e = 0.5(49) = 24.5 \text{ kpsi}$$

Next, we determine the Marin factors. For the machined surface, Eq. (6-19) with Table 6-2 gives

$$k_a = aS_{ut}^b = 2.70(49^{-0.265}) = 0.963$$

For axial loading, from Eq. (6-21), the size factor  $k_b = 1$ , and from Eq. (6-26) the loading factor is  $k_c = 0.85$ . The temperature factor  $k_d = 1$ , since we accounted for the temperature in modifying the ultimate strength and consequently the endurance limit. For 99 percent reliability, from Table 6-5,  $k_e = 0.814$ . Finally, since no other conditions were given, the miscellaneous factor is  $k_f = 1$ . The endurance limit for the part is estimated by Eq. (6-18) as

$$\begin{aligned} S_e &= k_a k_b k_c k_d k_e k_f S'_e \\ &= 0.963(1)(0.85)(1)(0.814)(1)24.5 = 16.3 \text{ kpsi} \end{aligned}$$

For the fatigue strength at 70 000 cycles we need to construct the  $S$ - $N$  equation. From p. 277, since  $S_{ut} = 49 < 70$  kpsi, then  $f = 0.9$ . From Eq. (6-14)

$$a = \frac{(fS_{ut})^2}{S_e} = \frac{[0.9(49)]^2}{16.3} = 119.3 \text{ kpsi}$$

and Eq. (6-15)

$$b = -\frac{1}{3} \log \left( \frac{f S_{ut}}{S_e} \right) = -\frac{1}{3} \log \left[ \frac{0.9(49)}{16.3} \right] = -0.1441$$

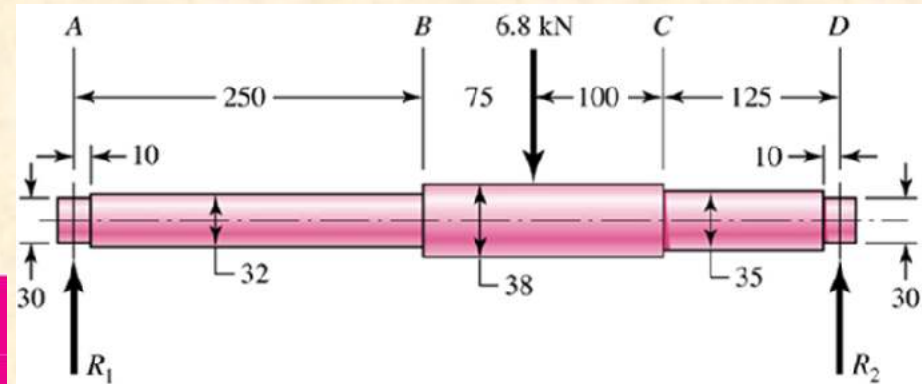
Finally, for the fatigue strength at 70 000 cycles, Eq. (6-13) gives

$$S_f = a N^b = 119.3(70\,000)^{-0.1441} = 23.9 \text{ kpsi}$$

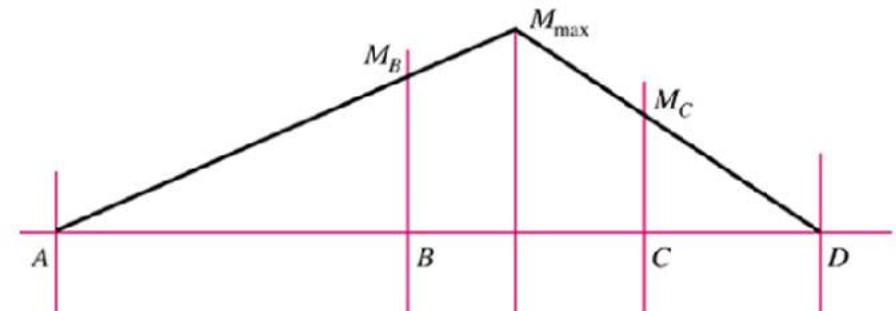
# EXAMPLE 6-9 (Textbook)

Figure 6-22a shows a rotating shaft simply supported in ball bearings at  $A$  and  $D$  and loaded by a nonrotating force  $F$  of 6.8 kN. Using ASTM “minimum” strengths, estimate the life of the part.

## SOLUTION



(a)

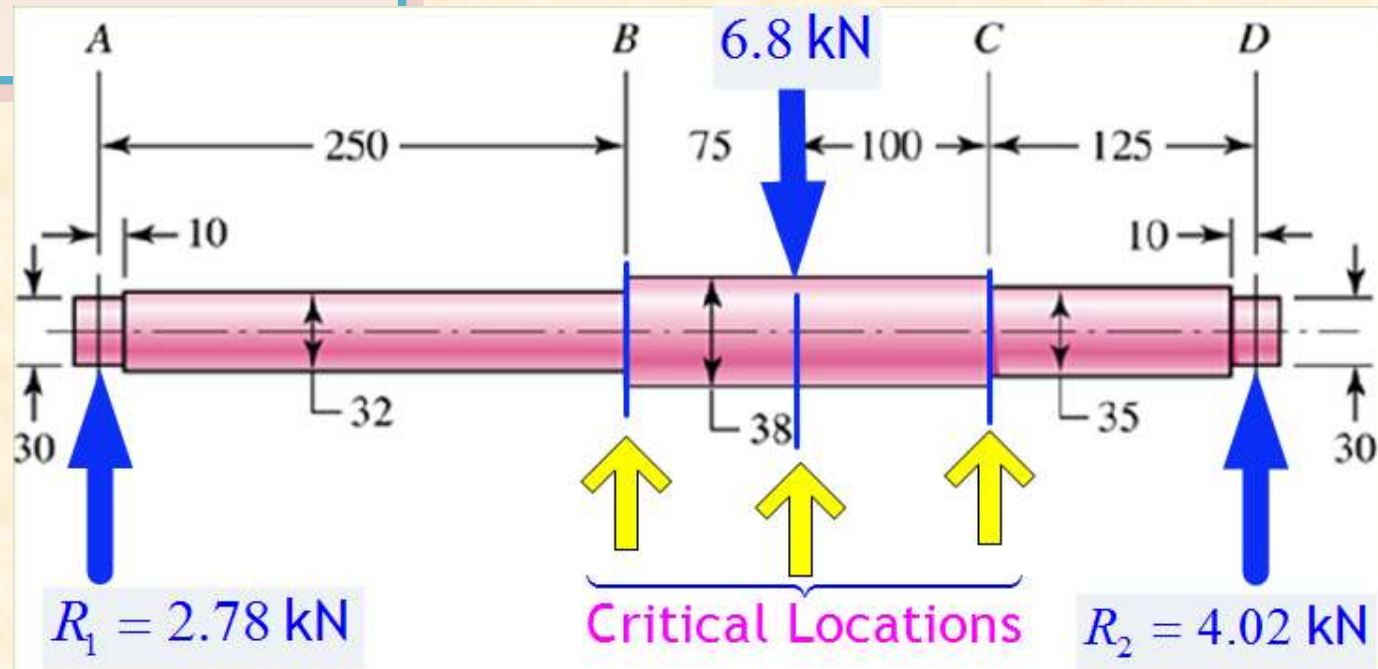


(b)

| 1       | 2                   | 3          | 4                            | 5                          |
|---------|---------------------|------------|------------------------------|----------------------------|
| UNS No. | SAE and/or AISI No. | Processing | Tensile Strength, MPa (kpsi) | Yield Strength, MPa (kpsi) |
| G10500  | 1050                | HR         | 620 (90)                     | 340 (49.5)                 |
|         |                     | <b>CD</b>  | <b>690 (100)</b>             | <b>580 (84)</b>            |



- **Failure** will probably occur at **B** rather than **C** or at the point of **maximum bending moment**.
- **Point B** has:
  - a smaller cross-section
  - a higher bending moment
  - a higher stress concentration factor than C.
- **Location** of the maximum bending moment has a **larger size** and **no stress concentration**.



We shall solve the problem by first estimating the strength at point  $B$ , since the strength will be different elsewhere, and comparing this strength with the stress at the same point.

From Table A-20 we find  $S_{ut} = 690$  MPa and  $S_y = 580$  MPa. The endurance limit  $S'_e$  is estimated as

$$S'_e = 0.5(690) = 345 \text{ MPa}$$

From Eq. (6-19) and Table 6-2,

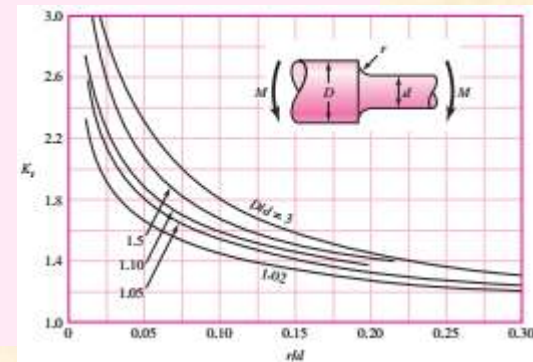
$$k_a = 4.51(690)^{-0.265} = 0.798$$

From Eq. (6-20),

$$k_b = (32/7.62)^{-0.107} = 0.858$$

Since  $k_c = k_d = k_e = k_f = 1$ ,

$$S_e = 0.798(0.858)345 = 236 \text{ MPa}$$



To find the geometric stress-concentration factor  $K_t$  we enter Fig. A-15-9 with  $D/d = 38/32 = 1.1875$  and  $r/d = 3/32 = 0.09375$  and read  $K_t \doteq 1.65$ . Substituting  $S_{ut} = 690/6.89 = 100$  kpsi into Eq. (6-35) yields  $\sqrt{a} = 0.0622 \sqrt{\text{in}} = 0.313 \sqrt{\text{mm}}$ . Substituting this into Eq. (6-33) gives

$$K_f = 1 + \frac{K_t - 1}{1 + \sqrt{a/r}} = 1 + \frac{1.65 - 1}{1 + 0.313/\sqrt{3}} = 1.55$$

$$K_f = 1 + \frac{K_t - 1}{1 + \sqrt{a/r}}$$

$$\text{Bending or axial: } \sqrt{a} = 0.246 - 3.08(10^{-3})S_{ut} + 1.51(10^{-5})S_{ut}^2 - 2.67(10^{-8})S_{ut}^3$$

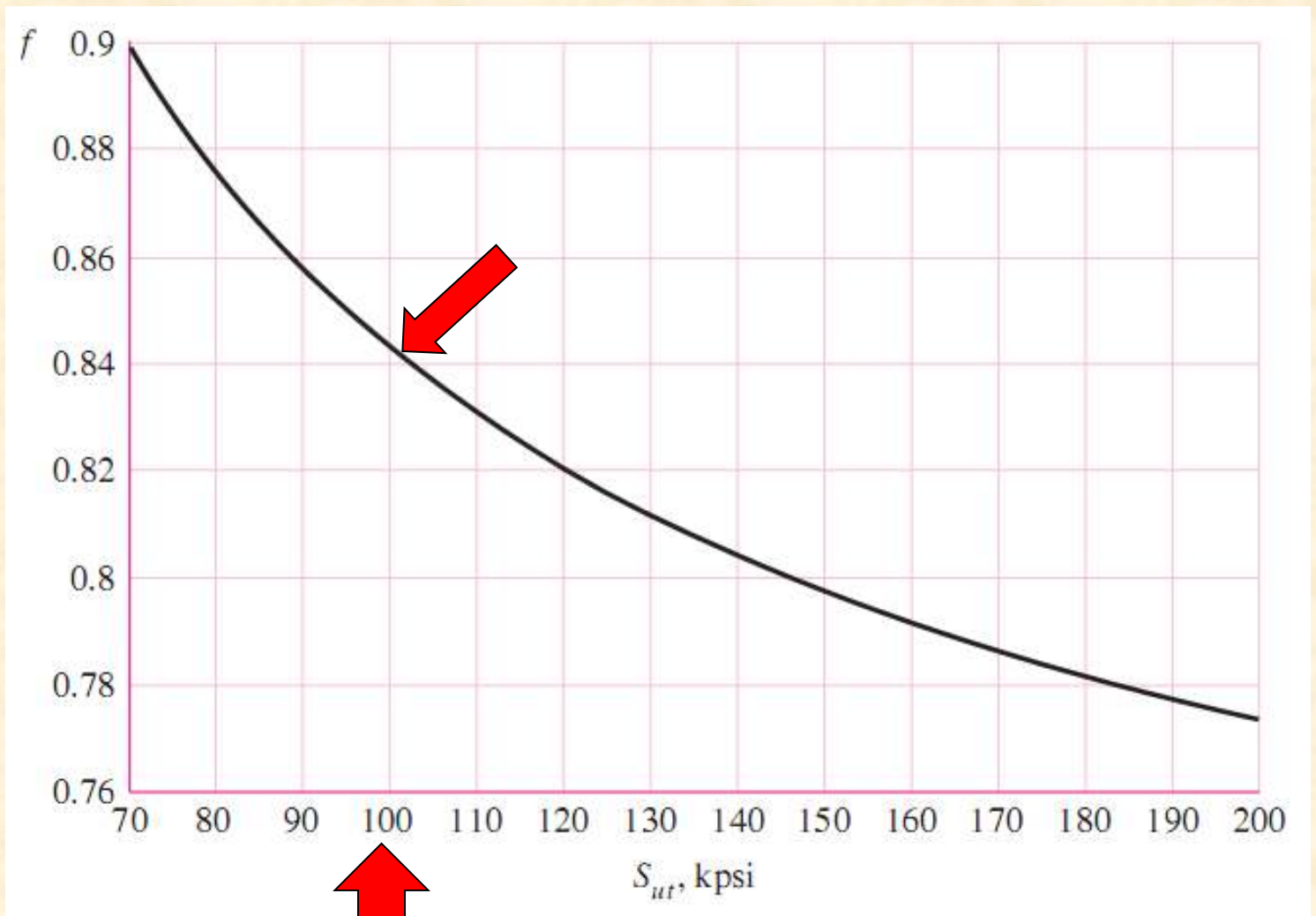
The next step is to estimate the bending stress at point  $B$ . The bending moment is

$$M_B = R_1 x = \frac{225F}{550} 250 = \frac{225(6.8)}{550} 250 = 695.5 \text{ N} \cdot \text{m}$$

Just to the left of  $B$  the section modulus is  $I/c = \pi d^3/32 = \pi 32^3/32 = 3.217 (10^3) \text{mm}^3$ . The reversing bending stress is, assuming infinite life,

$$\sigma = K_f \frac{M_B}{I/c} = 1.55 \frac{695.5}{3.217} (10)^{-6} = 335.1 (10^6) \text{Pa} = 335.1 \text{ MPa}$$

This stress is greater than  $S_e$  and less than  $S_y$ . This means we have both finite life and no yielding on the first cycle.



For finite life, we will need to use Eq. (6-16). The ultimate strength,  $S_{ut} = 690$  MPa = 100 kpsi. From Fig. 6-18,  $f = 0.844$ . From Eq. (6-14)

$$a = \frac{(f S_{ut})^2}{S_e} = \frac{[0.844(690)]^2}{236} = 1437 \text{ MPa}$$

and from Eq. (6-15)

$$b = -\frac{1}{3} \log \left( \frac{f S_{ut}}{S_e} \right) = -\frac{1}{3} \log \left[ \frac{0.844(690)}{236} \right] = -0.1308$$

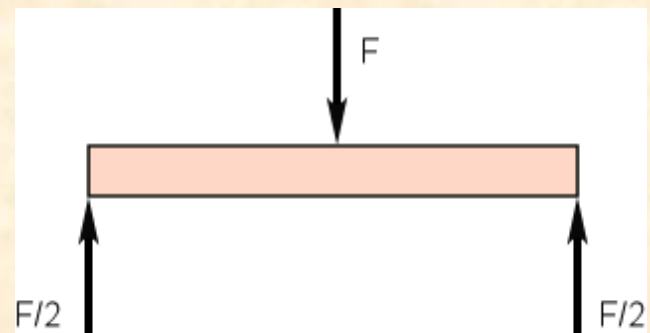
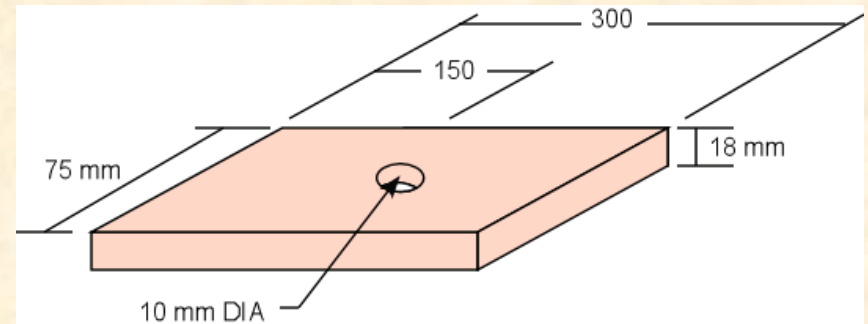
From Eq. (6-16),

$$N = \left( \frac{\sigma_a}{a} \right)^{1/b} = \left( \frac{335.1}{1437} \right)^{-1/0.1308} = 68(10^3) \text{ cycles}$$

# Example #04

- Given:

- $S_{ut} = 1400 \text{ MPa}$
- $S_{yt} = 950 \text{ MPa}$
- Heat-treated (as-forged)
- $F_{\text{mean}} = 9.36 \text{ kN}$
- $F_{\text{max}} = 10.67 \text{ kN}$
- $d/w = 0.133$ ;  $d/h = 0.55$

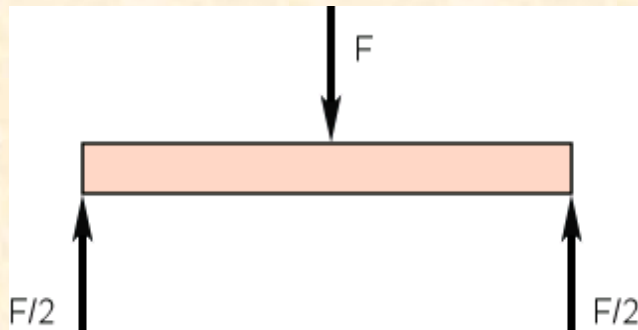
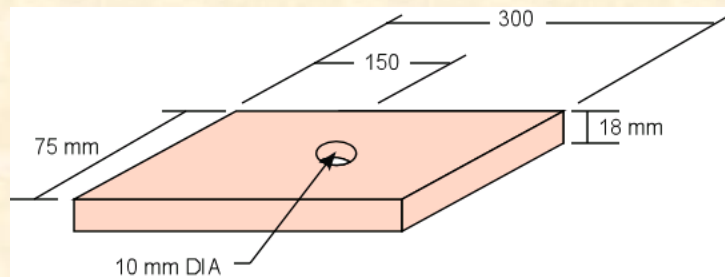


- Find:

- $N_{\text{cycle}}$  for finite life, or factor of safety for infinite life, assuming  $F_{\text{mean}}$  is constant

# Example, cont.

- Find  $\sigma_m$  and  $\sigma_a$



$$\sigma = \frac{My}{I}$$

$$I = \frac{1}{12}bh^3 = \frac{1}{12}(w-d)h^3 = \frac{1}{12}(75-10)18^3 = 3.16 \times 10^{-8} \text{ m}^4$$

$$y_{\max} = \frac{h}{2} = 0.009 \text{ m}$$

$$M_m = \left(\frac{F_m}{2}\right)\left(\frac{L}{2}\right) = \frac{1}{4}F_m L = \frac{1}{4}(9.36 \times 10^3)(0.3) = 702 \text{ Nm}$$

$$M_{\max} = \left(\frac{F_{\max}}{2}\right)\left(\frac{L}{2}\right) = \frac{1}{4}F_{\max} L = \frac{1}{4}(10.67 \times 10^3)(0.3) = 800 \text{ Nm}$$

$$\sigma_m = \frac{M_m y_{\max}}{I} = 200 \text{ MPa}$$

$$\sigma_{\max} = \frac{M_{\max} y_{\max}}{I} = 228 \text{ MPa}$$

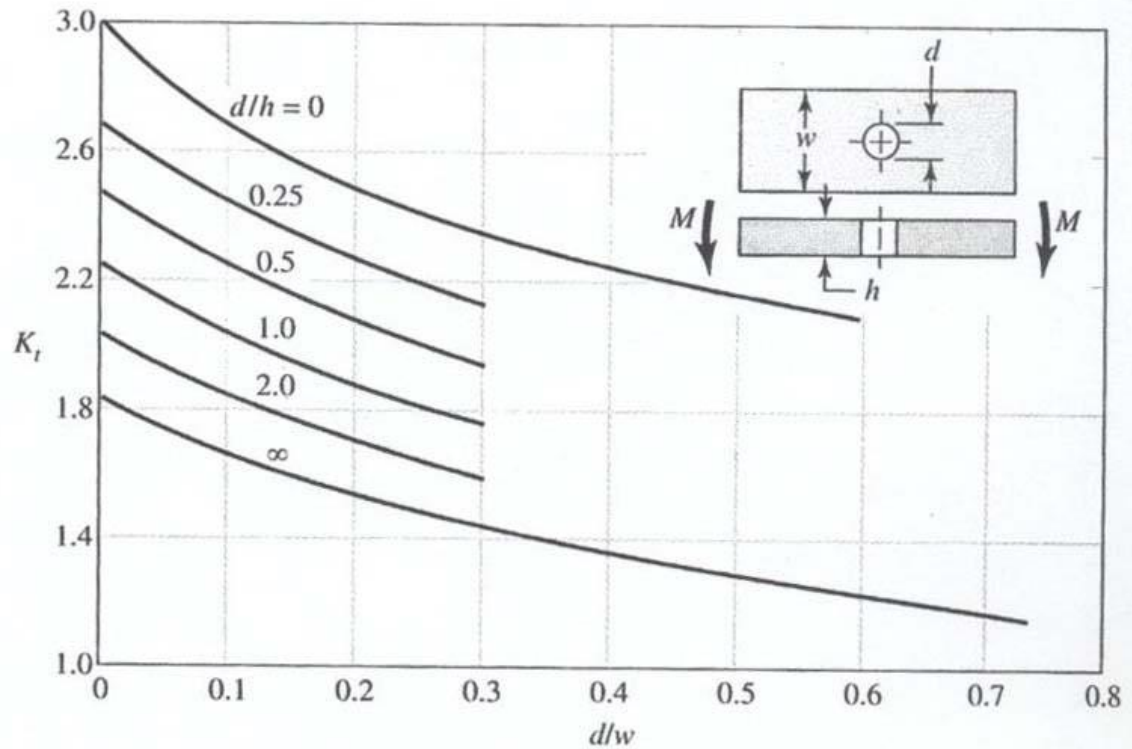
$$\sigma_a = \sigma_{\max} - \sigma_m = 28 \text{ MPa}$$

# Stress Concentration Factor

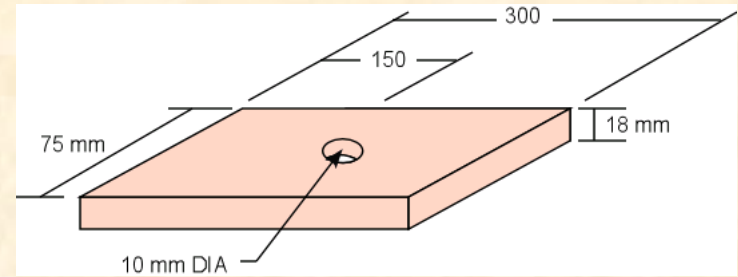
**Figure A-15-2**

Rectangular bar with a transverse hole in bending.

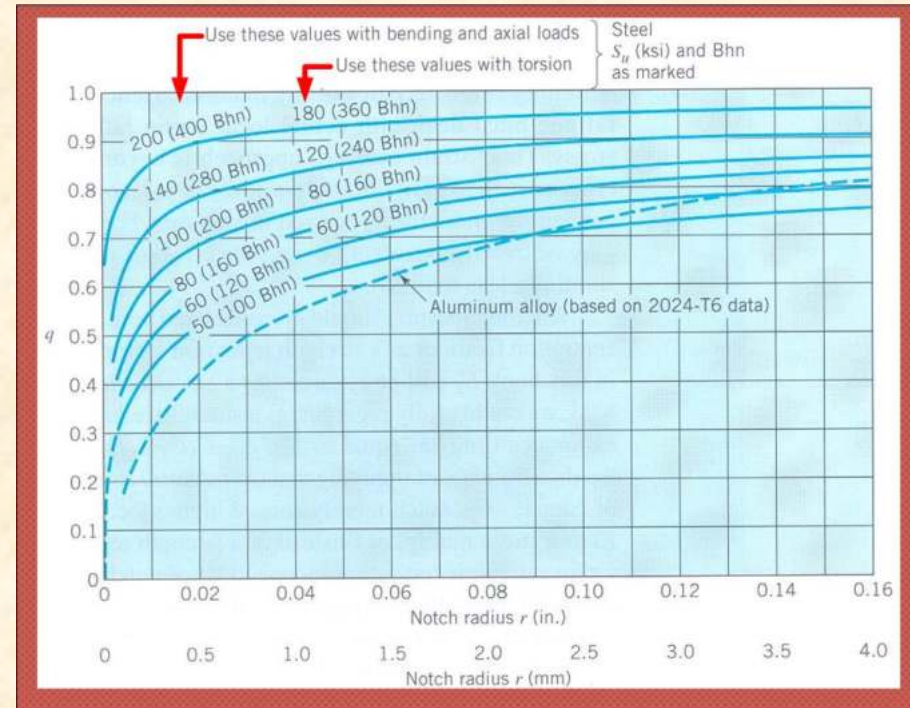
$\sigma_0 = Mc/I$ , where  
 $I = (w - d)h^3/12$ .



# Example, cont.



- Since this is uniaxial loading,
  - $\sigma_m = 200 \text{ MPa}$
  - $\sigma_a = 28 \text{ MPa}$
 } nominal
- We need to take care of the SC factors
  - $S_u = 1400 \text{ MPa}$
  - $k_f = 1 + q(k_t - 1)$
  - $k_t \sim 2.2$  (Figure A15-2)
  - $q \sim 0.95$  (Figure 6-20)
  - $k_f = 2.14$

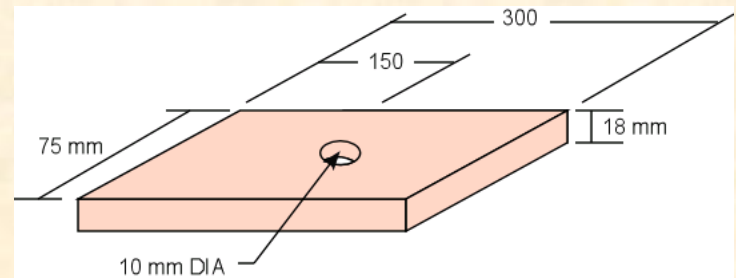


$$\sigma_a = \sigma'_a = k_f \sigma_{a_{nom}} = (2.14)(28) = 60 \text{ MPa}$$

$$\sigma_m = \sigma'_m = k_f \sigma_{m_{nom}} = (2.14)(200) = 428 \text{ MPa}$$

# Example, cont.

- Find strength
- Eqn. 6-8:  $S'_e = .5 \times S_{ut}$   
 $S'_e \sim 700 \text{ MPa}$  since  $S_{ut} = 1400 \text{ MPa}$
- Modification factors



Surface :

$$k_a = aS_{ut}^b$$

$$a = 271$$

$$b = -0.995$$

$$k_a = 0.201$$

Size :

Equation (7-19):

$$2.8 \leq d_{eq} \leq 51 \text{ mm}$$

$$d_{eq} = 0.808(hb)^{\frac{1}{2}}$$

$$k_b = 1.24d_{eq}^{-0.107}$$

$$k_b = 0.86$$

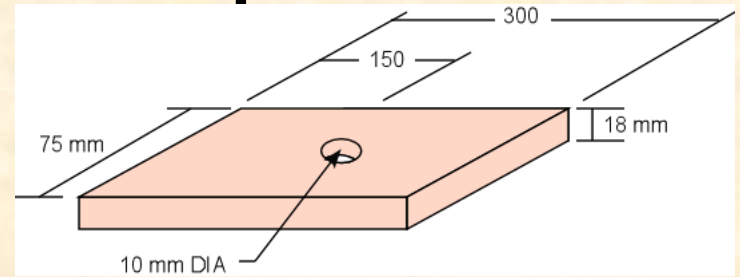
Load:

Bending

$$k_c = 1 \text{ (Eq. 7-25)}$$

$$S_e = (0.201)(0.86)(700) = 121 \text{ MPa}$$

# Example,



- Design criteria
  - Goodman line:

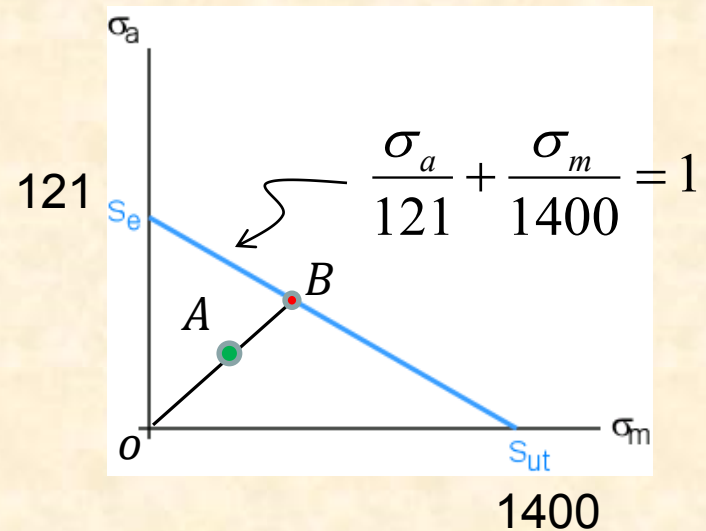
$$\frac{\sigma_a}{S_e} + \frac{\sigma_m}{S_{ut}} = 1/n$$

- For arbitrary variation in  $\sigma_a$  and  $\sigma_m$ ,

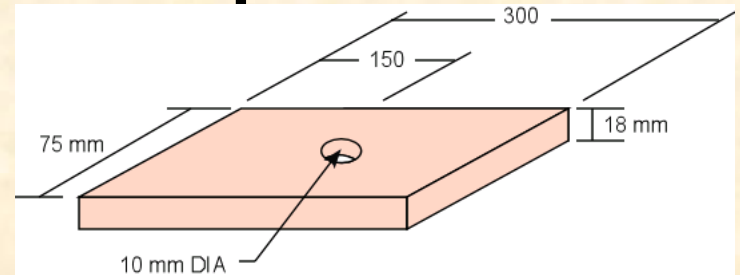
$$\frac{\sigma_a}{121} + \frac{\sigma_m}{1400} = \frac{1}{\eta}$$

$$\frac{1}{\eta} = \frac{60}{121} + \frac{428}{1400}$$

$$\eta = 1.25$$



# Example,



- However, we know that  $F_{\text{mean}} = \text{constant}$  from problem statement

□  $\sigma_m = \text{constant}$

$$\frac{S_a}{S_e} + \frac{\sigma_m}{S_{ut}} = 1$$

$$\frac{S_a}{121} + \frac{428}{1400} = 1$$

$$S_a = 84 \text{ MPa}$$

$$\eta = \frac{S_a}{\sigma_a} = \frac{84}{60} = 1.4 \quad \text{Less conservative!}$$

